

Design and calibration of a cryogenic blackbody calibrator at centimeter wavelengths

A. Kogut^{a)} and E. Wollack

NASA/Goddard Space Flight Center, Laboratory for Astronomy and Solar Physics, Greenbelt, Maryland 20771

D. J. Fixsen, M. Limon, and P. Mirel

SSAI Code 685, NASA/GSFC Laboratory for Astronomy and Solar Physics, Greenbelt, Maryland 20771

S. Levin and M. Seiffert

Jet Propulsion Laboratory, California Institute of Technology, 4800 Oak Grove Drive, Pasadena, California 91109

P. M. Lubin

Department of Physics, University of California at Santa Barbara, Santa Barbara, California 93106

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We describe the design and calibration of an external cryogenic blackbody calibrator used for the first two flights of the Absolute Radiometer for Cosmology, Astrophysics, and Diffuse Emission (ARCADE) instrument. The calibrator consists of a microwave absorber weakly coupled to a superfluid liquid helium bath. Half-wave corrugations viewed 30° off axis reduce the return loss below -35 dB within a compact footprint. Ruthenium oxide resistive thermometers embedded within the absorber monitor the temperature across the face of the calibrator. The thermal calibration transfers the calibration of a reference thermometer to the flight thermometers using the flight thermometer readout system. The calibrator thermometry is stable in time over four years, with statistical uncertainty in the temperature calibration of order 2 mK near 2.7 K, limited primarily by thermal fluctuations in the liquid helium bath. Observations of the superfluid transition demonstrate that the absolute temperature scale is accurate within 0.3 mK. © 2004 American Institute of Physics. [DOI: 10.1063/1.1821622]

I. INTRODUCTION

The Absolute Radiometer for Cosmology, Astrophysics, and Diffuse Emission (ARCADE) is a balloon-borne instrument designed to measure the temperature of the cosmic microwave background (CMB) at centimeter wavelengths.¹ ARCADE uses a set of narrow-band cryogenic radiometers to compare the sky to an external blackbody calibration target, in order to detect or limit deviations from a blackbody spectrum. At centimeter wavelengths, raw sensitivity is not an important design criterion; the instrument is designed instead to reduce or eliminate major sources of systematic uncertainty. The instrument is fully cryogenic; all major components are independently temperature controlled to remain near 2.7 K, isothermal with the signal from deep space. Boiloff helium vapor, vented through the aperture plane, forms a barrier between the instrument and the atmosphere at 35 km altitude; there are no windows or warm optics to correct. An independently controlled blackbody calibrator located on the antenna aperture plane rotates to cover each antenna in turn, so that each antenna alternately views the sky or a known blackbody. The cryogenic design minimizes the effect of internal reflection or absorption: any residual instrumental signals cancel to first order in the sky-calibrator

comparison. All radiometers view the same calibrator in turn, eliminating cross-calibration uncertainties when comparing the sky temperature between different frequency channels.

The external calibrator is critical to the ARCADE experiment. It establishes a common blackbody reference for all frequency channels, and establishes an absolute temperature reference for comparison with other instruments. Assuring the electromagnetic performance of the calibrator, though nontrivial, is straightforward. The ability to achieve ARCADE's science goals can thus be traced to precision and accuracy with which the physical temperature distribution within the calibrator can be monitored.

II. CALIBRATOR DESIGN

The experimental design places multiple constraints on the calibrator. It must maintain stable cryogenic temperatures while located at the mouth of an open liquid helium bucket Dewar, and must be compact enough to avoid spillover from the side lobes of antennas viewing the sky. The absorber should be nearly isothermal, so that a limited number of temperature sensors can sample the temperature distribution across the calibrator with sufficient precision to provide the necessary radiometric calibration.

Figure 1 shows the schematic design of the calibration target. It consists of a microwave absorber (Emerson and Cuming Eccosorb CR-112, an iron-loaded epoxy) cast with

^{a)}Author to whom correspondence should be addressed; electronic mail: alan.j.kogut@nasa.gov

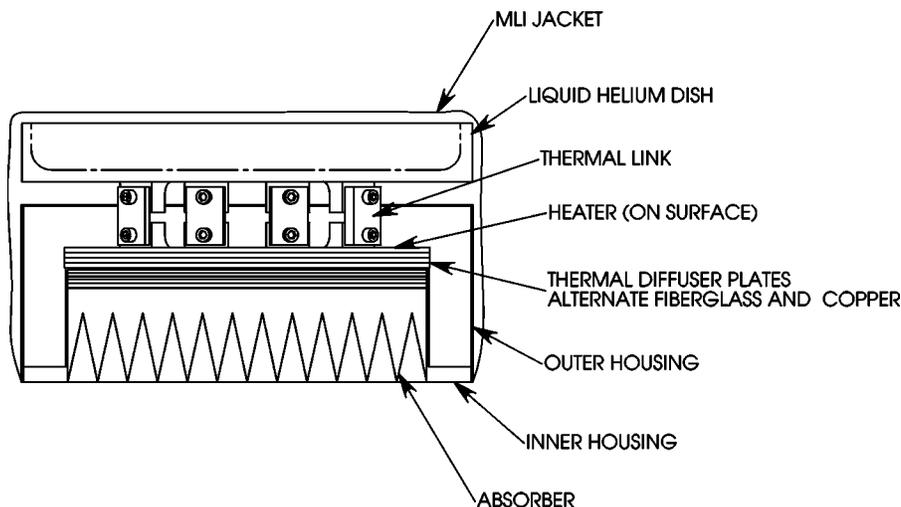


FIG. 1. Schematic of ARCADE external blackbody calibrator. The microwave absorber (Eccosorb CR-112) is weakly coupled to a superfluid liquid helium reservoir through a series of copper/fiberglass buffer plates. The absorber is 21 cm across, with corrugated front surface to reduce reflected power.

grooves in the front surface to reduce surface reflections. To increase viscosity and reduce settling of the iron powder during the curing process, we added 0.5% cabosil powder by mass to the Eccosorb prior to casting. Cabosil, a fused silica powder, has absorption coefficient much less than the unloaded Eccosorb. Similar cabosil loadings of an Eccosorb absorber have been used in calibrators at higher frequencies; its presence in small quantities does not otherwise affect the absorption or reflection properties of the Eccosorb.^{2,3} Absorber samples prepared at GSFC, including the cabosil filler, have complex permittivity $\epsilon'_r=3.9$ and $\epsilon''_r=0.5$, for a loss tangent $\tan \delta=0.13$ measured at 30 GHz. The Eccosorb is mounted on a set of two thermally conductive oxygen-free copper plates separated by fiberglass thermal insulators, each 1.6 mm in thickness. Thermal control is achieved by heating the outermost copper plate, which is in weak thermal contact with a superfluid helium reservoir. Radial thermal gradients at each stage are reduced by the ratio of the thermal conductance of the buffer plates. A two-stage design is sufficient to reduce the thermal “footprint” of the heater resistors well below 1 mK at the Eccosorb/copper interface.

The calibrator is located at the mouth of an open bucket Dewar and must maintain temperatures near 2.7 K despite heating from the thermal control loop, infrared emission from the atmosphere, and residual condensation of atmospheric nitrogen. A stainless steel shell protects the calibrator sides and top, with a 20-layer reflective blanket providing additional isolation from the environment. Fountain-effect pumps transport superfluid liquid helium from the main Dewar to a separate reservoir inside the calibrator, providing the necessary cooling without requiring moving parts. Boiloff gas from the calibrator reservoir vents through holes in the top cover plate.

Eccosorb CR-112 has a high index of refraction ($n \approx 4$), producing a power reflection coefficient $R > 0.25$ at normal incidence for frequencies below 30 GHz.² At millimeter wavelengths, blackbody calibrators typically minimize reflections using deep conical designs to ensure multiple reflections.^{3,4} At longer wavelengths, this becomes impractical due to size constraints. The front Eccosorb surface is cast instead into corrugated grooves. The ARCADE antennas view the calibrator at an angle $\theta=30^\circ$ from normal

incidence.¹ In the limit of a monochromatic incident plane wave, the spacing of the corrugations can be chosen so that reflections from one corrugation are exactly out of phase with the neighboring corrugations, cancelling the $m=0$ “specular” reflection from the calibrator for all odd multiples of the incident wavelength. Although the electric field at the antenna aperture is not a plane wave, the analysis provides some guidance to minimize reflections within a compact geometry. Yokimori⁵ has examined the effect of changing the pitch/height ratio for corrugated dielectric surfaces and finds minimal reflection for corrugation height 1–2 times the pitch. The ARCADE calibrator uses corrugations 3 cm deep spaced 1.5 cm apart, minimizing reflections from the Eccosorb surface in both the 10 and 30 GHz channels. An additional 1.3 cm of absorber between the bottom of the corrugations and the copper mounting structure provides additional absorption to reduce reflections from the metal backing.

III. MICROWAVE PERFORMANCE

Calibration by an external target corrects reflection and attenuation within the instrument to first order, with the notable exception of the external calibrator reflectivity. Reflections from the calibrator create a systematic offset in the sky-calibrator comparison proportional to the calibrator’s power reflection coefficient and the temperature difference between the calibrator and the sky. During flight, the calibrator will be commanded to different temperatures that could vary from the sky temperature by as much as several K. Achieving 1 mK or better accuracy for the sky temperature thus requires calibrator power reflection below -30 dB.

Figure 2 shows the power reflection coefficient measured for both ARCADE frequency bands using gated time-of-flight techniques at room temperature with the antenna viewing the calibrator in flight configuration (30° from normal incidence). The maximum in-band reflection is -43 dB at 10 GHz, with gain-weighted average -49 dB across the band. At 30 GHz the maximum reflection is -28 dB, with gain-weighted average -35 dB. The results do not depend significantly on the orientation of the absorber corrugations to the antenna. Measurements of the antenna/calibrator coupling using a vector network analyzer provide similar limits

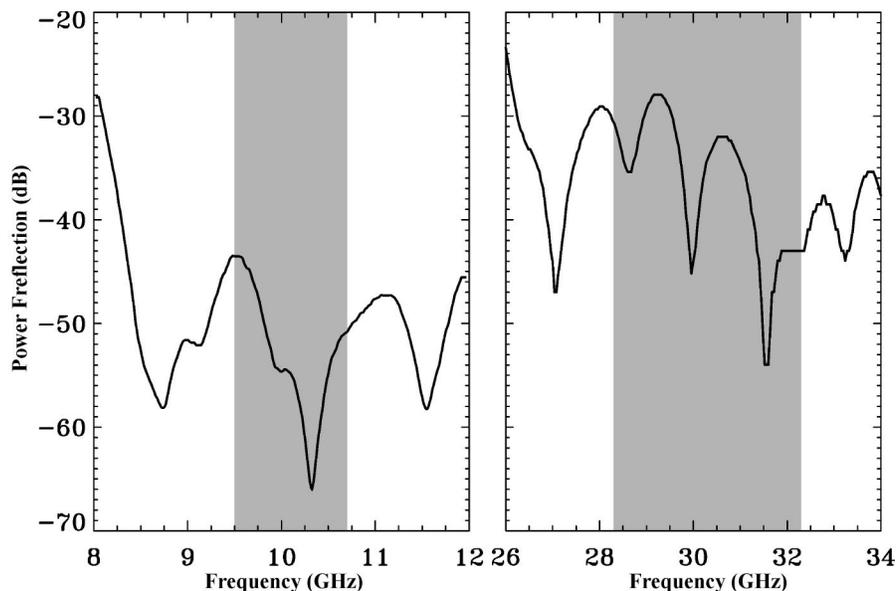


FIG. 2. Power reflection coefficient measured using ARCADE calibrator and antennas in the flight off-axis configuration. The gray band shows the passband at each frequency range. The gain-averaged in-band reflection is below -35 dB in each band.

$R < -35$ dB at 30 GHz. As the absorber cools, the epoxy filler shrinks and the conductivity of the iron will increase, increasing the reflectivity and decreasing the absorption. Measurements suggest an increase in power reflection of 30%–60% at cryogenic temperatures.² Both bands still easily meet the -30 dB requirement.

The cryogenic absorber within the calibrator is backed by a reflective copper plate and completely covers the antenna aperture without windows, infrared blocking filters, or other emissive elements between the absorber and the antenna.¹ The calibrator emissivity is thus determined by the measured reflection coefficient without correction from additional elements in the beam. Leakage around the target is small. To allow mechanical movement, the calibrator is held 5 mm above the antenna aperture. Radiation leaking through the resulting gap is measured to be less than -50 dB. The measured reflection coefficients thus translate to gain-weighted emissivity 0.999 99 at 10 GHz and 0.9997 at 30 GHz.

IV. THERMOMETRY

We monitor the calibrator temperature using ruthenium oxide resistance thermometers, chosen for their small size, stability, and large resistance change at cryogenic temperatures. The thermometers consist of commercially available thick-film chip resistors (Dale Electronic RCWP-550) with $76 \mu\text{m}$ manganin leads providing four-wire resistance measurements. The resistance at room temperature is $10 \text{ k}\Omega$, rising to $25 \text{ k}\Omega$ with slope $dT/dR = -0.158 \text{ mK}/\Omega$ at 2.7 K . Five thermometers embedded within the microwave absorber monitor the temperature distribution, while an additional two thermometers mounted next to the heater on the outermost copper plate provide feedback for thermal control. Figure 3 shows the location of the thermometers in the calibrator. Three thermometers (labeled T1, T2, and T3) are embedded near the tips of the Eccosorb corrugations, with two other thermometers (T4 and T5) located in the absorber midway between the bottom of the corrugations and the copper mounting plate. Thermometers T2 and T5 are located atop

each other at the center of the calibrator to provide an estimate of vertical gradients within the absorber. After fabrication, the entire calibrator, including the embedded thermometers, was cycled ten times between 300 and 77 K to relieve thermal stress. A sixth thermometer embedded within the absorber failed to open during this initial thermal cycling. The calibrator has since been cooled repeatedly below 2 K with no adverse mechanical or electrical effects.

We calibrate the thermometers *in situ* by immersing the entire calibrator in a liquid helium bath and monitoring the resistance of each thermometer while slowly lowering the pressure above the bath. A NIST-traceable reference thermometer records the bath temperature using the same read-

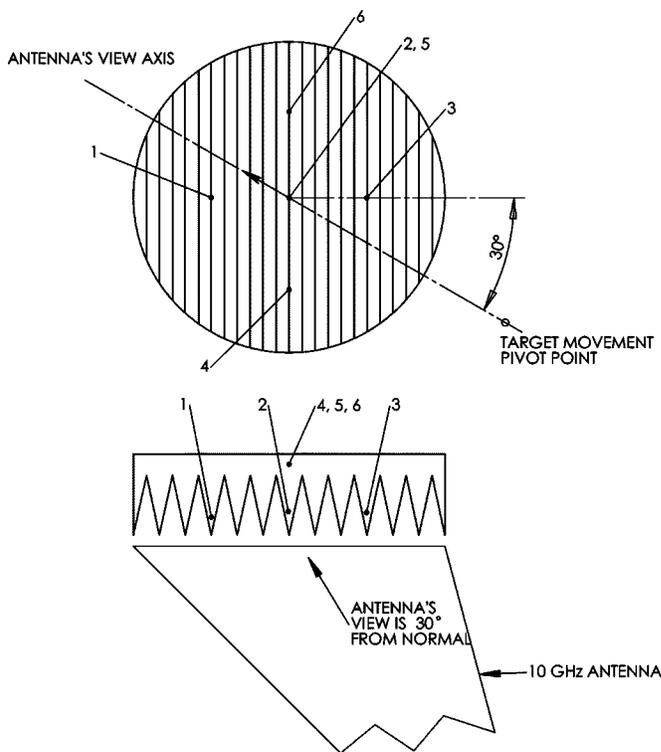


FIG. 3. Location of thermometers within the calibrator.

out as the calibrator thermometers. Calibrations four years apart using two different readout systems agree within the statistical uncertainties (~ 0.3 mK). The initial calibration in 1998 used a $1 \mu\text{A}$ excitation current sinusoidally modulated at frequency 1 Hz. A lockin circuit demodulates the voltage drop across each thermometer and integrates for 5.6 s to yield the thermometer resistance. The 1998 calibration sequentially measured the five embedded thermometers plus the reference bath thermometer. Each thermometer was thus sampled once every 33.4 s as the bath temperature fell from 4.2 to 1.4 K over the course of 10 h. To prevent bath stratification from producing systematic thermal gradients in the Dewar, we use only data taken when the bath temperature was falling fast enough to ensure vigorous boiling. We assume that the bath and calibrator are isothermal and interpolate the measured bath temperature to the time recorded for each calibrator resistance measurement.

We calibrated the thermometers a second time in 2002 using the flight thermometer readout system.⁶ The 2002 calibration differed from the 1998 calibration in several important aspects. The 2002 calibration used a faster square-wave modulation at 75 Hz, applying $1 \mu\text{A}$ excitation current to each thermometer for only 26.7 ms and multiplexing to sample each thermometer once every 1.067 s. The synchronously demodulated voltage across the thermometer is thus of order 25 mV, providing detection and digitization with measurement uncertainty of order 1Ω .⁶ A set of four reference resistors monitored any drifts in the readout. Since we used the same system for ground calibration as flight, effects such as self-heating or the shunt impedance from stray capacitance in the leads are automatically accounted for. The 2002 calibration also included the two control thermometers mounted on the copper heater plate.

We obtained three calibration runs in 1998 and five more in 2002. Direct comparison between years requires a correction for self-heating. Although self-heating in the 2002 calibration is the same as in flight, it is not the same as the 1998 calibration which applied power to each thermometer for a much longer duration. Additional tests in 1998 were designed to measure the change in thermometer temperature as the excitation current varied from 1 to $10 \mu\text{A}$ at a series of bath temperatures ranging from 1.6 to 4.2 K. The resulting self-heating is well described by

$$\Delta T = \Delta T_0 \left(\frac{I}{1 \mu\text{A}} \right)^2 \left(\frac{T}{1 \text{K}} \right)^{-2.2}, \quad (1)$$

where $\Delta T_0 \sim 27$ mK (varying slightly among the thermometers) and the fitted power law in temperature includes the dependence of both the thermometer resistance and the thermal conductance of the absorber. We correct the 1998 data for self-heating but do not explicitly correct the 2002 data (or flight data) since the effect is included in both the ground and flight readout.

We interpolate the measured resistance data to generate a resistance to temperature calibration for each thermometer over the temperature range 1.4–4.2 K. The calibrations derived from each of the eight individual runs are in good agreement with each other. We quantify this by generating a set of “standard” resistance values spanning the observed

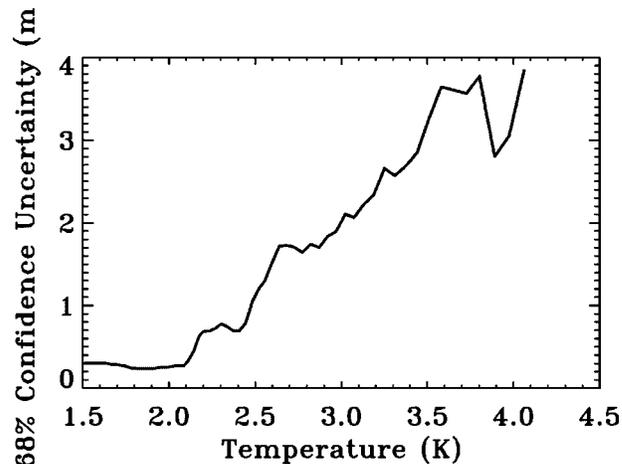


FIG. 4. Statistical uncertainty in the derived temperature calibration based on scatter from eight calibration runs in 1998 and 2002. The uncertainty at temperatures above the superfluid transition is dominated by thermal gradients in the bath.

range, then converting these to the equivalent temperature using the resistance-temperature calibration curves from each of the eight separate calibration runs. The resulting scatter in the computed temperatures at each resistance shows the agreement from run to run. Figure 4 shows the 68% confidence uncertainty in the mean for thermometer T1 as a function of bath temperature. Below the superfluid transition all calibrations agree within 0.3 mK, consistent with the noise of the thermometer readout.⁶ Above the transition temperature the uncertainty increases linearly with the bath temperature. This excess noise is not observed at colder temperatures, suggesting an origin from thermal fluctuations within the liquid helium bath itself.

We use the mean of the eight individual calibration runs to produce a single resistance-temperature calibration for each thermometer. Observations of the superfluid helium transition at temperature 2.1768 K provide an independent check on the absolute calibration. We record the resistance of each thermometer during each run as the pressure above the bath slowly falls. The superfluid transition produces a distinctive kink in the time-ordered data as the thermal properties of the bath abruptly change. Figure 5 shows all observations of the superfluid transition for thermometer T1 from the 1998 and 2002 calibration runs. The calibrated temperatures correctly reproduce the known transition temperature.

Additional data in 2003 provide thermometer calibration at temperatures from 4.2 to 20 K. This higher temperature calibration placed the calibrator inside a copper cryostat mounted within a liquid helium bucket Dewar. We used the flight thermometer readout system to measure the resistance of each flight thermometer as the liquid level in the dewar fell below the calibrator enclosure. The absorber temperature rose from 4.2 to 20 K over a period of 20 h, much longer than the few second time constant of the calibrator, ensuring near isothermality across the absorber. Data taken with the absorber immersed in liquid helium agreed with previous measurements from the 1998 and 2002 calibrations.

If the calibrator was isothermal, its only contribution to the uncertainty in the CMB temperature would be the abso-

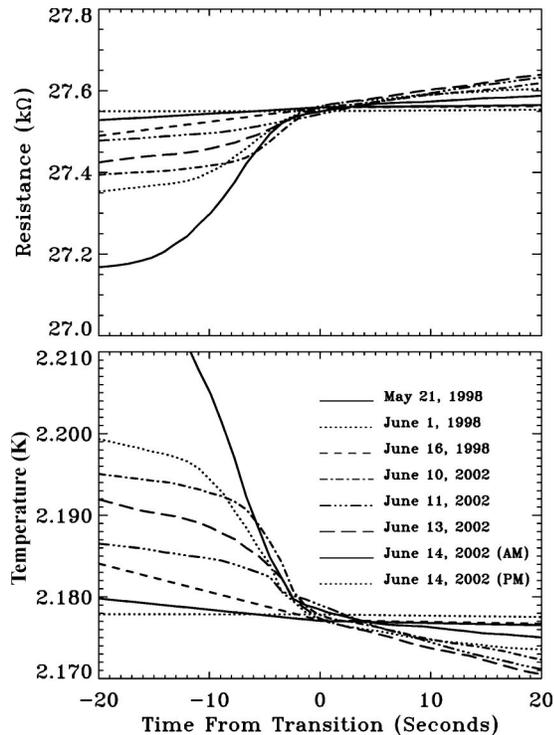


FIG. 5. Resistance and calibrated temperatures for calibrator thermometer T1 from all calibration runs, as the bath temperature passes through the superfluid transition. The superfluid transition creates a distinctive knee in the time-ordered data. (top) Measured resistance values. (bottom) Calibrated temperatures: the calibrated temperature for each run correctly reproduces the transition at $T=2.1768$ K.

lute calibration uncertainty of the embedded thermometers. Spatial gradients within the Eccosorb will alter the relation between the physical temperature measured at the thermometers and the microwave power measured by the radiometer. We embed five thermometers within the microwave absorber at widely spaced intervals and at different heights relative to the absorber tips, allowing direct measurement of both radial and back-to-front vertical gradients within the absorber (Fig. 3). The power received by the radiometer is the volume integral of the temperature distribution within the absorber,

weighted by the electric field distribution at the antenna aperture. Provided that the spatial scale for gradients is comparable to or larger than interthermometer separation, the integral may be approximated as a linear combination of the temperatures measured by the five embedded thermometers. We may test for gradients on smaller length scales (localized hot or cold spots that would not be adequately sampled by the thermometers) by using data from the 2003 flight.⁷ We compute the radiometric temperature of the target using a best-fit linear combination of either three, four, or all five thermometers. Unsampled hot or cold spots must contribute to the radiometric temperature, but would not be included in the linear combination. The scatter in the resulting CMB temperature as successive thermometers are dropped from the fit thus serves as an estimate of residual uncertainties from unsampled gradients. Details are provided in a companion paper.⁷ We find that the measured CMB temperature is stable as successive thermometer are dropped from the fit, demonstrating that the five embedded thermometers provide adequate spatial sampling of thermal gradients within the calibrator.

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¹A. Kogut, E. Wollack, D. Fixsen, M. Limon, P. Mirel, S. Levin, M. Seiffert, and P. Lubin, *Astrophys. J., Suppl. Ser.* **154**, 493 (2004).

²H. Hemmati, J. Mather, and W. Eichhorn, *Rev. Sci. Instrum.* **24**, 4489 (1985).

³J. Mather, D. Fixsen, R. Shafer, C. Mosier, and D. Wilkinson, *Astrophys. J.* **512**, 511 (1999).

⁴H. Gush, M. Halpern, and E. Wishnow, *Phys. Rev. Lett.* **65**, 537 (1990).

⁵K. Yokimori, *Appl. Opt.* **23**, 2303 (1984).

⁶D. Fixsen, P. Mirel, A. Kogut, and M. Seiffert, *Rev. Sci. Instrum.* **73**, 3659 (2002).

⁷D. J. Fixsen *et al.*, *Astrophys. J.* **611**, 86 (2004).